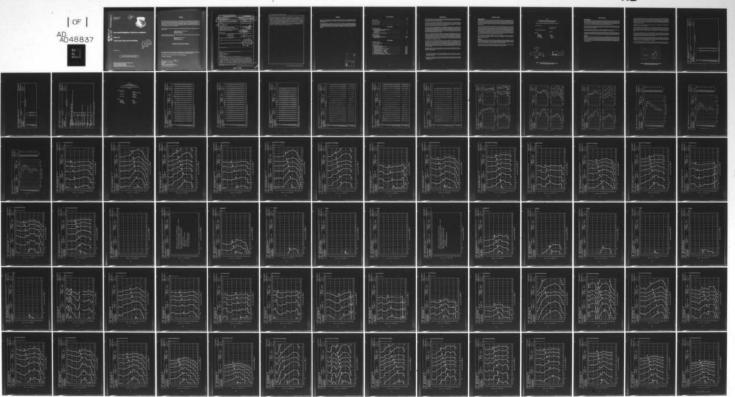
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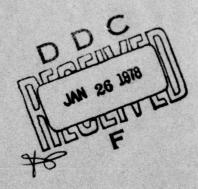




USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

Volume 79

T-28D Aircraft, Near and Far-Field Noise



FEBRUARY 1977

Approved for public release; distribution unlimited.

AEROSPACE MEDICAL RESEARCH LABORATORY AEROSPACE MEDICAL DIVISION AIR FORCE SYSTEMS COMMAND WRIGHT-PATTERSON AIR FORCE BASE, OHIO 45433

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C.m Cui

FOR THE COMMANDER

HENNING E. VON GIERKE

Director

Biodynamics and Bionics Division Aerospace Medical Research Laboratory

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Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 50-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distances from the source. Refer to Volume of this handbook, PUSAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application, AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The author gratefully acknowledges Mr. John Cole for his assistance in preparing this report, Mr. Robert England for his assistance in acquiring the raw data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing, and Mrs. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

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INTRODUCTION

The USAF T-28D is a trainer-type aircraft powered by an R1820-86A reciprocating engine. The aircraft was manufactured by the Columbus Division of North American Rockwell and the engines by the Wright Aeronautical Division of Curtis-Wright.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the T-28D aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15°C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 2 and 3) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise, AMRL-TR-75-50 (2), AMRL, WPAFB, OH, 1975.

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the T-28D aircraft during ground runup operations of its reciprocating engine. For these tests the aircraft was located on a concrete runup pad at Hurlburt Field, Eglin, AFB, with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the engine condition. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all the noise samples on magnetic tape. During analysis of each sample, he determined the root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the two near-field locations where ground crews are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations are difficult since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the T-28D aircraft at the two ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS FOR NEAR-FIELD NOISE MEASUREMENTS

T-28D Aircraft, Ground Runup, Hurlburt Field, Eglin AFB, 6 Aug 1971

Ground Crew Location

•

Engine Start, Fire Guard

9

Wheel Chock Pull

Aircraft Engine (and AGE) Operation

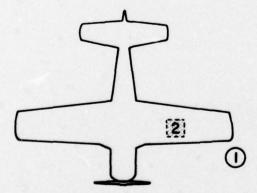
A

Idle Power

Meteorology

Temperature
Bar Pressure
Rel Humidity
Wind — Speed
— Direction

28.9 C 0.763 M Hg 72% 1 M/Sec (2 kt) 60 Deg



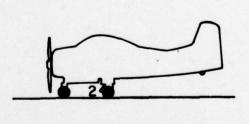


Figure 1. Near-Field Measurement Locations on Runup Pad at Hurlburt Field, Eglin AFB FL

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired both near- and far-field data during a 1-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the ground runup pad, ground cover, aircraft orientation and the 19 microphone measurement sites on the semicircle. The center of the 30 meter radius semicircle used in surveying the R1820-86A engine was on the ground directly below the intersection of the aircraft's centerline and the plane passing through the engine's propeller plane.

Table 4 provides cockpit readouts of some engines characteristics (RPM and manifold pressure) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All microphone measurement sites are in the acoustic far-field of the source where the sound wavefronts spherically diverge and the noise source may be regarded as a point source.

Test personnel acquired far-field noise data at Hurlburt Field by using a hand-held microphone (1.7 meters/5½ feet above the ground plane and pointed at the noise source, 0° incidence) and sequentially recording 5-10 seconds of data at each far-field location on a portable microphone/tape recorder system.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15°C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the T-28D aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

Estimates of noise levels for intermediate power conditions (e.g., 1800 RPM) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are, respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented at the 170 and/or 180 degree locations for highest power setting because of turbulent air flow behind the aircraft. Typically, the A-weighted levels for these angles are 10 to 20 dBA below those at the 160 degree microphone location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low (e.g., Table 5 and Figure 11 at idle power).

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

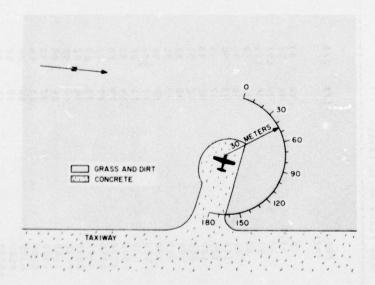


Figure 2. Far-Field Measurement Locations on Runup Pad at Hurlburt Field, Eglin AFB FL

2 1/3 OCTAVE BAND) ONEGA 3.2
NOISE SOURCE/SUBJECT!	-	OPERATIONS	() TEST 71-019-107
T-2AD ATROPAFT				1 04 060 74
GROUND CREW				
NEAR FIELD NOISE LEVELS	-		•) PAGE F1
		רסכי	LOCATION/CONDITION	
FRED	1/4	27.4		
(HZ)				
52	98	200		
31.5	98	95		
0.4	96	101		
50	16	101		
63	93	96		
00	06	16		
100	93	96		
125	87	16		
160	85	E 6		
250	200	16		
315	82	91		
00,	84	06		
200	7.8	86		
630	81	87		
900	96	91		
1000	79	88		
1250	11	06		
1600	62	89		
2000	91	92		
2500	85	06		
3150	*	0 0		
0000	100	200		
	0 0	0 0		
	* *	600		
10000	85	91		
OVERALL	102	108		

TABLE: MEASURED SOUND	PRESSUR	SOUND PRESSURE LEVEL (08) ND) IDENTIFICATION:) OMEGA 3.2
NOISE SOURCE/SUBJECT: 7-28D AIRCRAFT GROUND CREW NEAR FIELD NOISE LEVELS		C OPERATION:) TEST 71-019-107) RUN 01) 04 DEC 74) PAGE J1
			LOCATION/CONDITION	
FREQ (HZ)	1/A	2/A		
31.5	96	102		
63	94	103		
250	96	95		
200	98	93		
1000	9 0	95		
0004	6 6	63.		
8000	89	95		
OVERALL	102	108		

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15				
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	•	•	•	
NEAR FIELD NOISE LEVELS		•) PAGE H	H1
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	76 82			
	9			
AMERICAN OPTICAL 1700 EAR HUFF	S			
DASLA	73 78			
V-51R EAR PLUGS				
	096			
AMERICAN OPTICAL 1700 EAR MUFFS	MUFFS PLUS V-51R EAR PLUGS	S		
	•			
H-133 GROUNB COMMUNICATION UNI	-			
OASLA*	69 75		A TREE CONTRACTOR	
	6			
	460 M (2892) WOOD			
SPEECH	BE 94			
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SPEECH NOISE ECTION	960 960 INTERFERENCE LEVEL (PSIL IN DB) 86 94 EVEL, TONE CORRECTED (PNLT IN P C IN DB) 112 117	NOB)		

TABLE 4

TEST CONDITIONS FOR FAR-FIELD NOISE MEASUREMENTS

T-28D Aircraft, Ground Runup, Hurlburt Field, Eglin AFB 6 Aug 1971

Aircraft Engine Operation

Idle/Taxi Power

1200 RPM

19 Inches Manifold Pressure

Magneto Check

2250 RPM

30 RPM MAP

Military Power

2650 RPM

45 RPM MAP

Meteorology

Temperature
Bar Pressure
Rel Humidity
Wind — Speed
— Direction

28.9 C 0.763 M Hg 72 % 1 M/Sec (2 kt) 60 Deg

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72 70 67 66 67
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< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

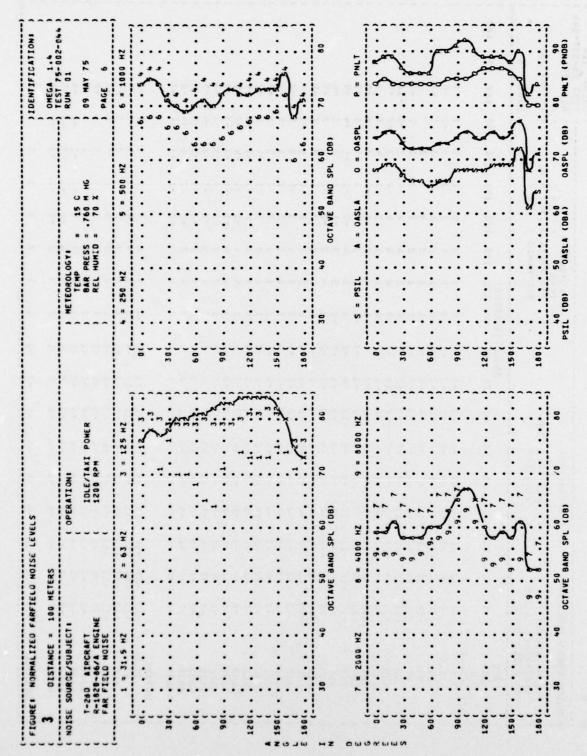
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160	100		66	96	66	101	101	102	105	106	101	108	108	109	107	106	107	101	88
200	91		91	90	88	88	88	93	96	16	96	16	46	100	66	66	16	90	81
250	91		06	88	91	88	69	46	95	16	95	93	96	96	96	*	95	94	81
315	35		93	91	91	88	95	95	91	91	96	*	95	95	96	97	83	81	18
004	16		93	95	91	06	91	91	93	93	95	96	16	96	6	96	8	*	91
200	93		91	80	68	98	68	91	93	36	93	93	96	96	93	3	80	81	43
630	60		60	20	20	90	10	92	98	60	60	60	95	86	66	06	92	8	6.
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1250	9.0	8	8	87		0 6	8		8	0 0	0	6	9	16	6 6	6 6	8 0	81	100
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OVERALL	107	105	101	106	101	106	106	108	111	113	115	115	116	116	114	112	109	103	46

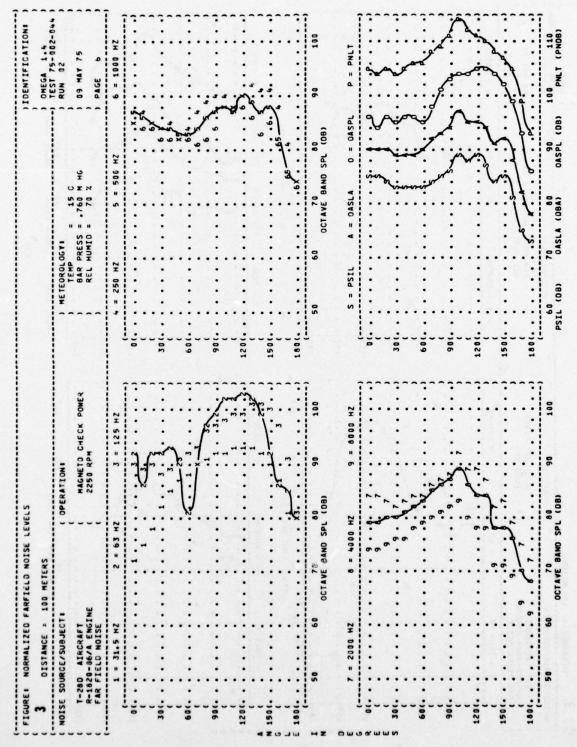
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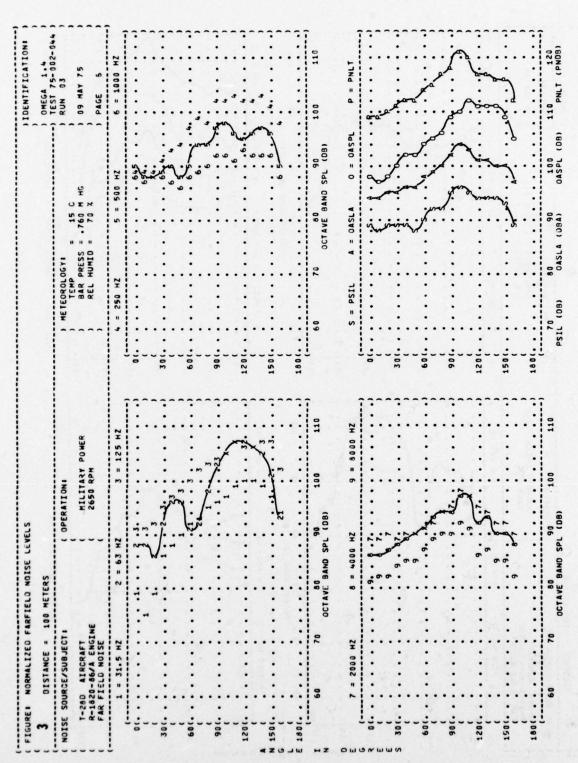
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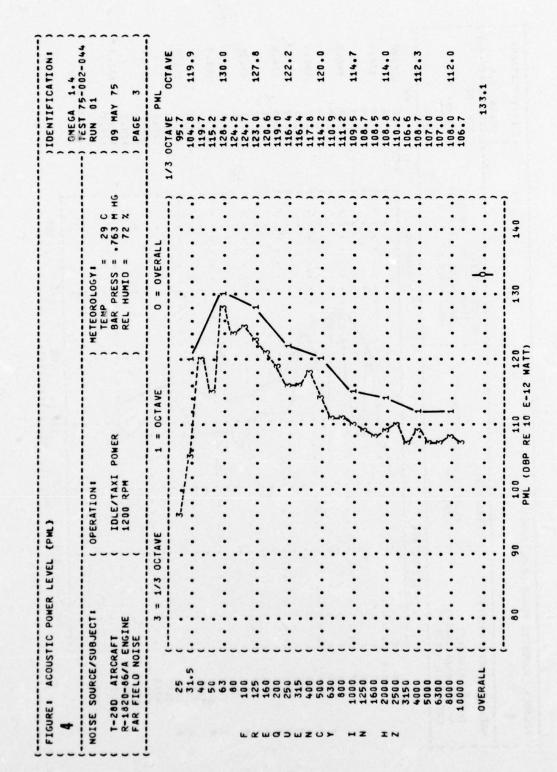
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FAR FIELD NO	OISE			_						-				-		^	PAGE	4	
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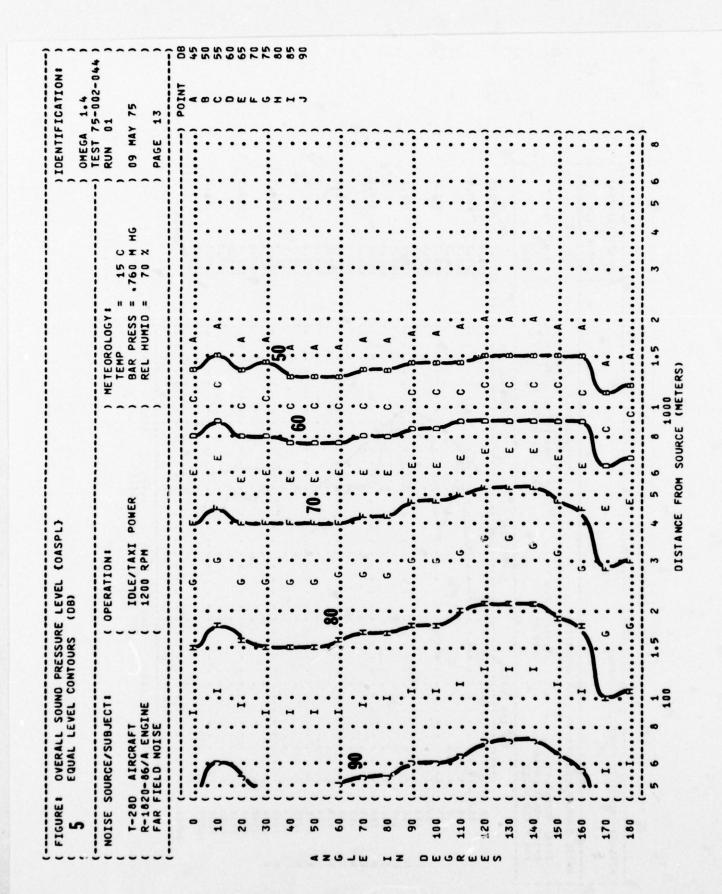


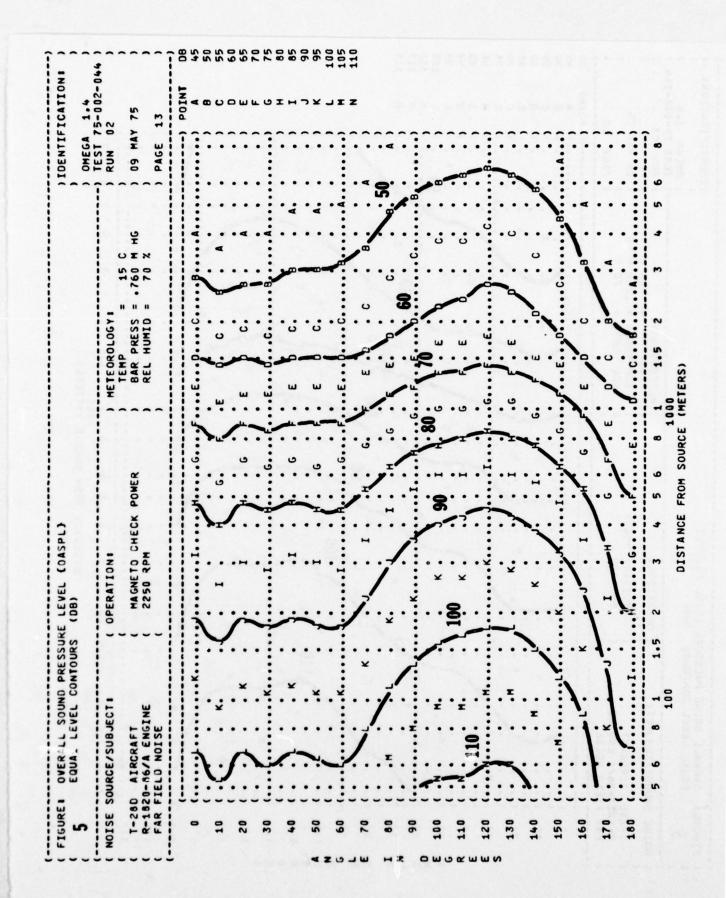


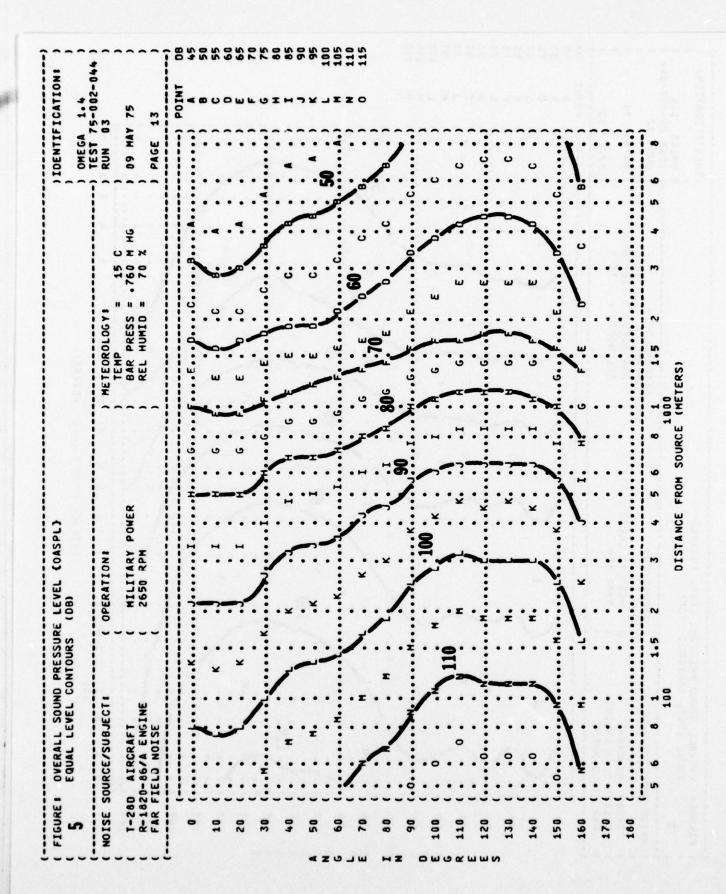


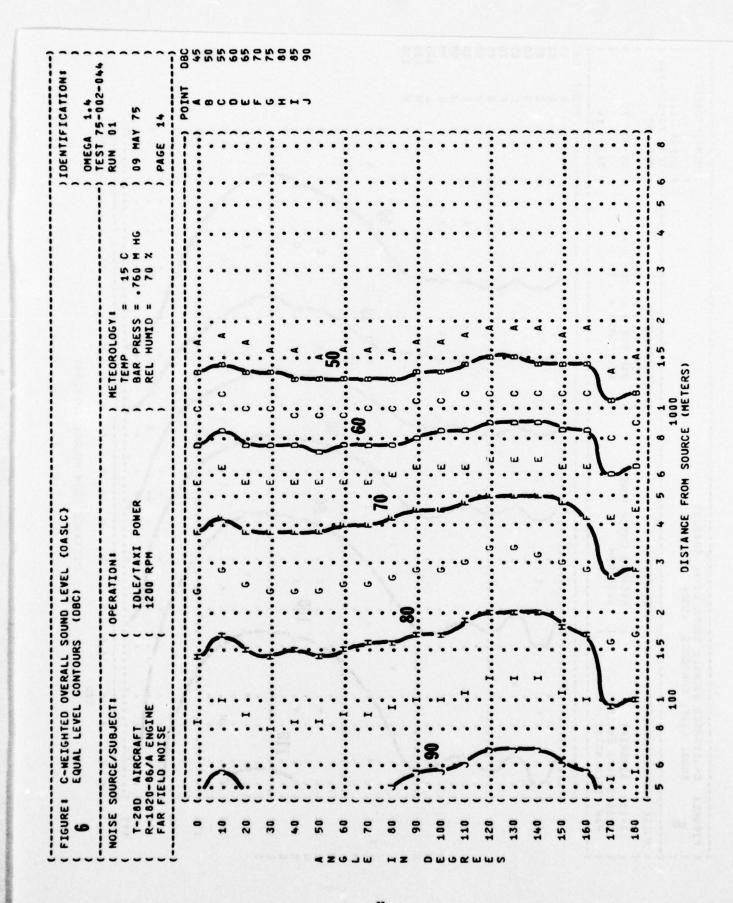
SE SOURCE/SUBJECT! (OPERATION!) NETEOROLOGY! 1-20 AIGCRAFT -1-20 AIGCRAFT -				
T-28D AIRCRAFT *-1820-86/A ENGINE *-1820-86/	ISE SOURCE/SUBJECT!	COPERATIONS) TEST 75-002-044
AR FIELD NOISE 3 = 1/3 OCTAVE 1 = OCTAVE	T-28D AIRCRAFT R-1820-86/A ENGINE	MAGNETO CHECK POWER	PRESS = .763 H HUMID = 72 %) 09 MAY 75
3 = 1/3 OCTAVE 1 = OCTAVE 0 = OVERALL 1/3 OCTAVE 121.9 5	FAR FIELD NOISE) PAGE 3
173 OVERALL (173 O	3 = 1/3	1 =	"	PWL
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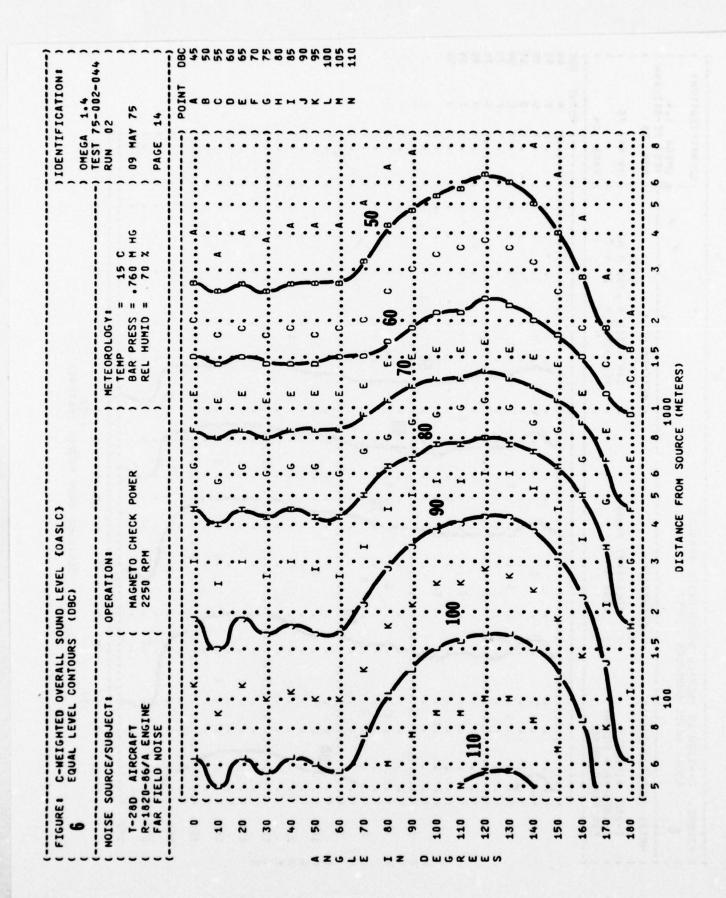
NOISE SOURCE/SUBJECT: (OPERATION: T-280 AIRCRAFT (HILITARY POWER R-1820-86/A ENGINE (2650 RPM FAR FIELD NOISE (2650 RPM 3 = 1/3 OCTAVE 1 25 (31.5 (32.	MER 1 = 0CTAVE) HE	TEOROLOGY: 29 C BAR PRESS = 763 H HG REL HUMID = 72 %) TEST 75-002-044) RUN 03) 09 HAY 75) PAGE 3
= 1/3 OCTAVE	= OCTAVE	O = OVERALL	PAGE 3 PML OCTAVE 0 123.5 143.9 136.3
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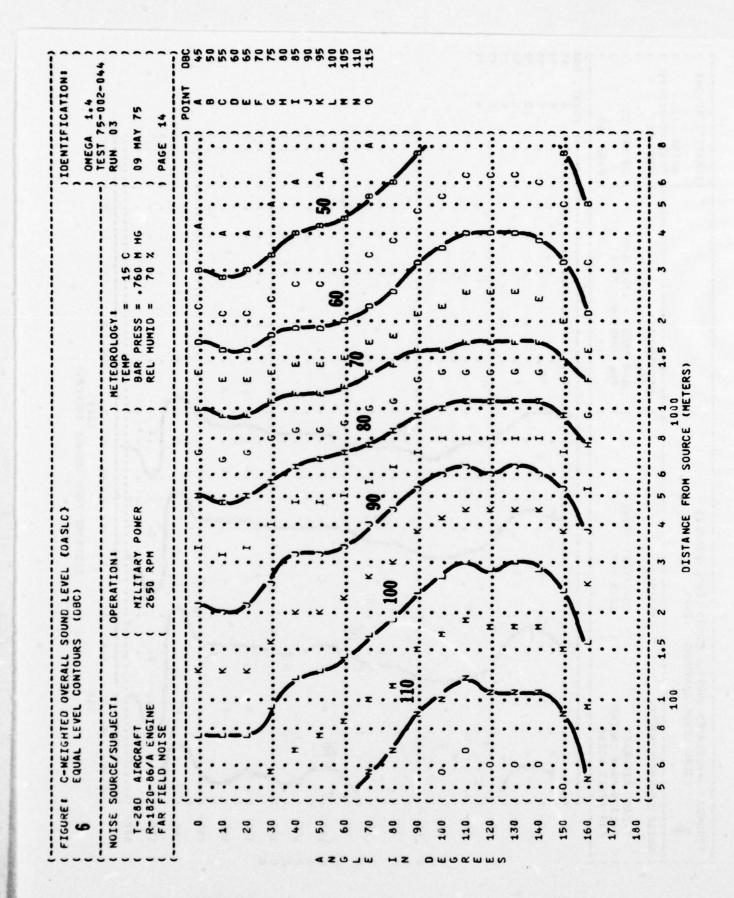


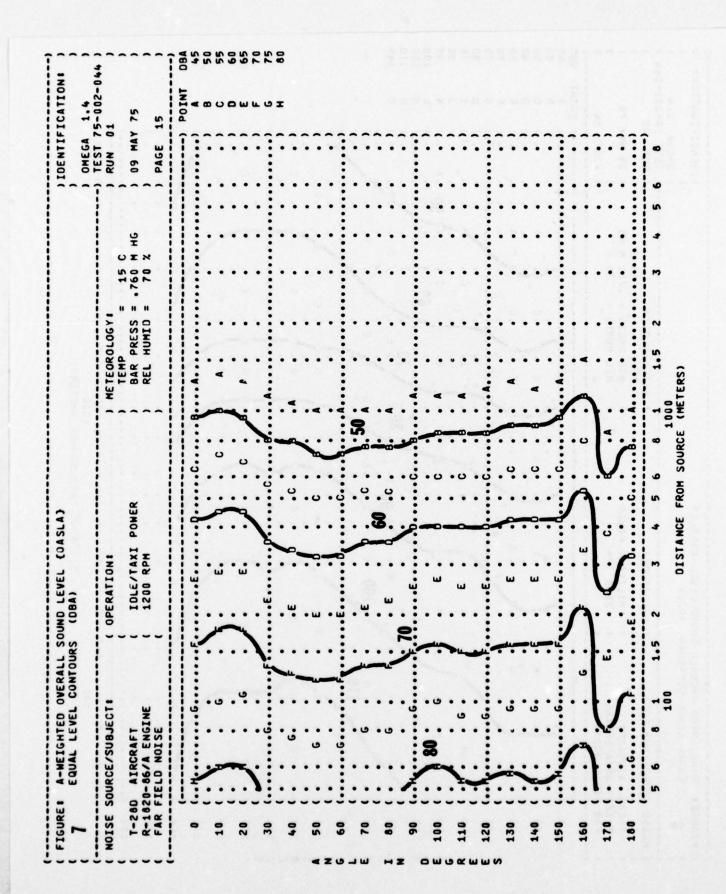


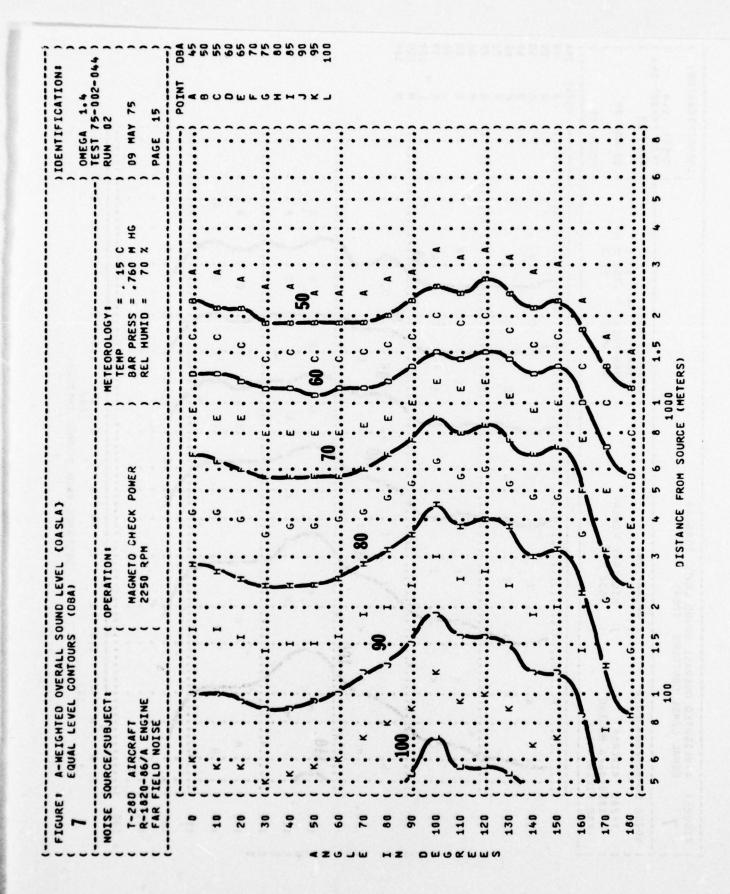


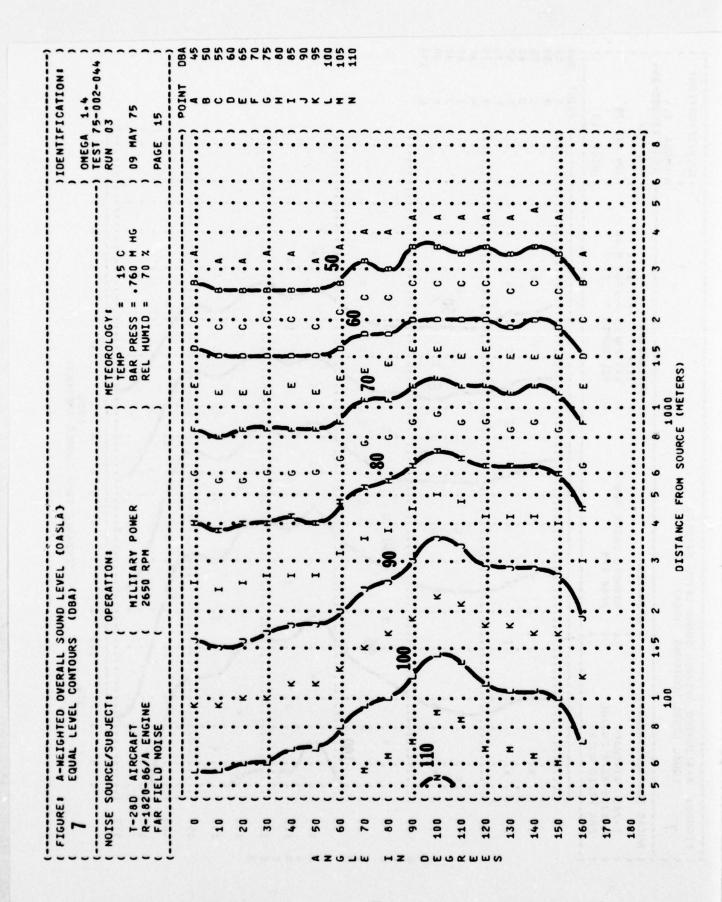


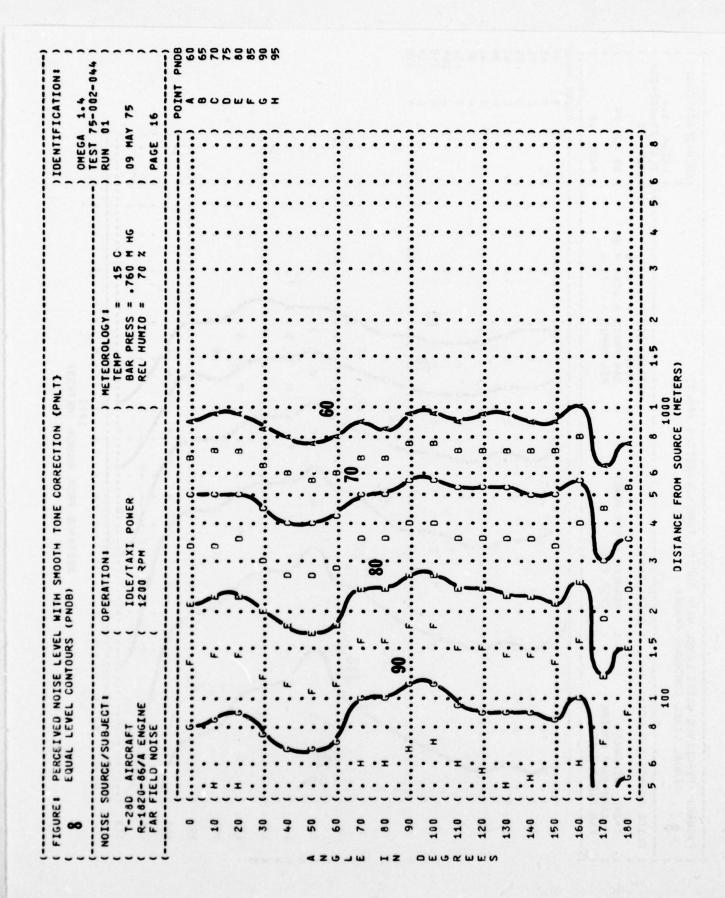


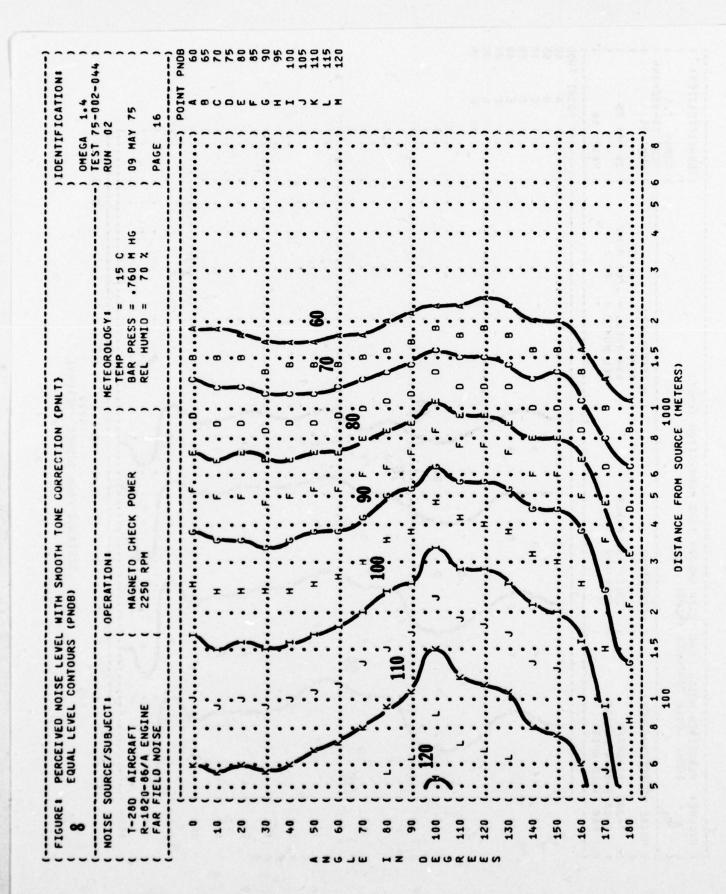


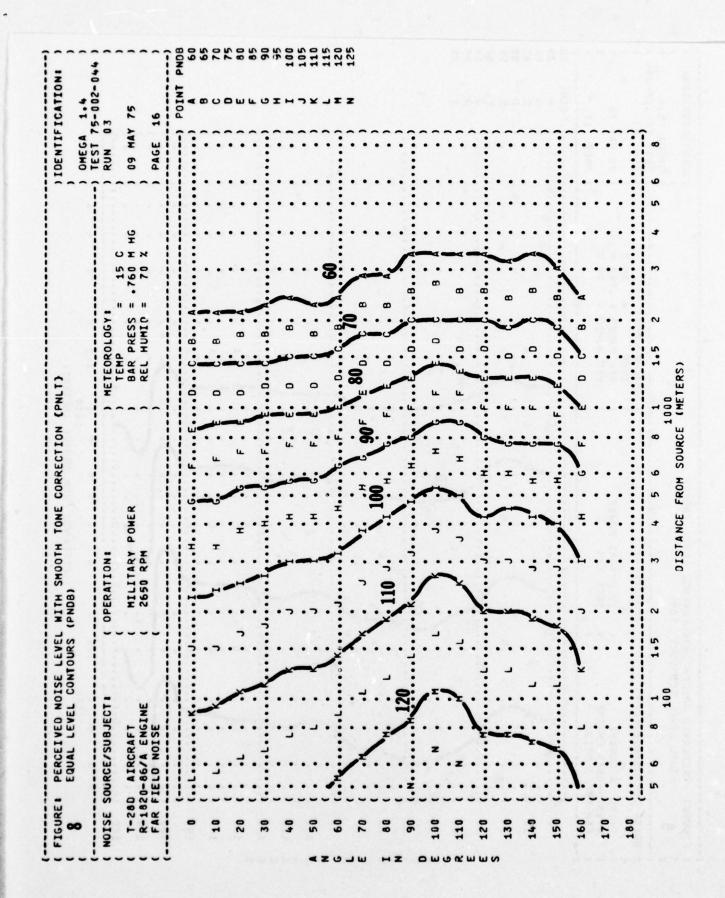


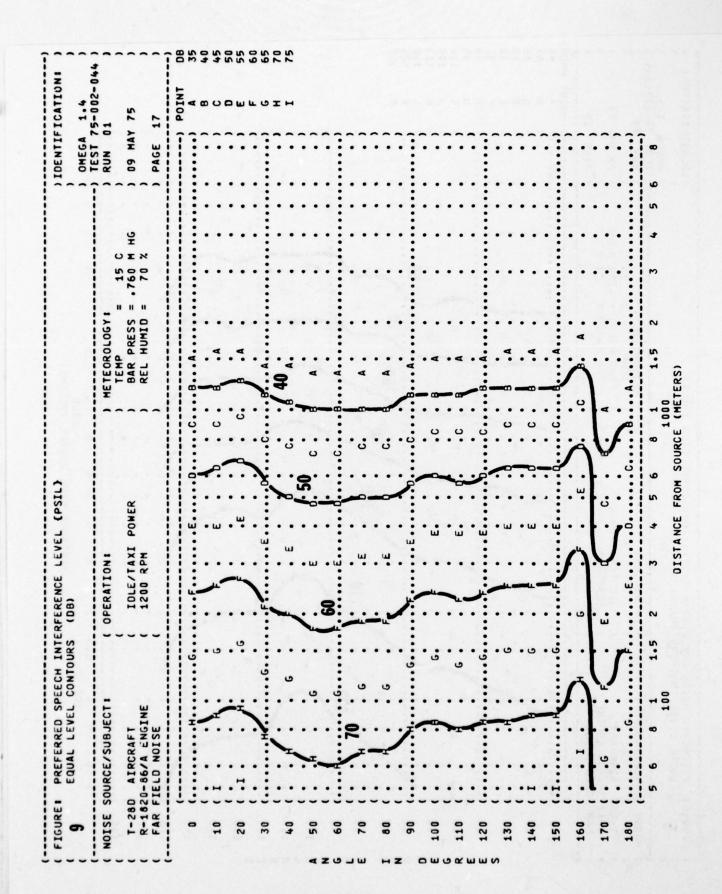


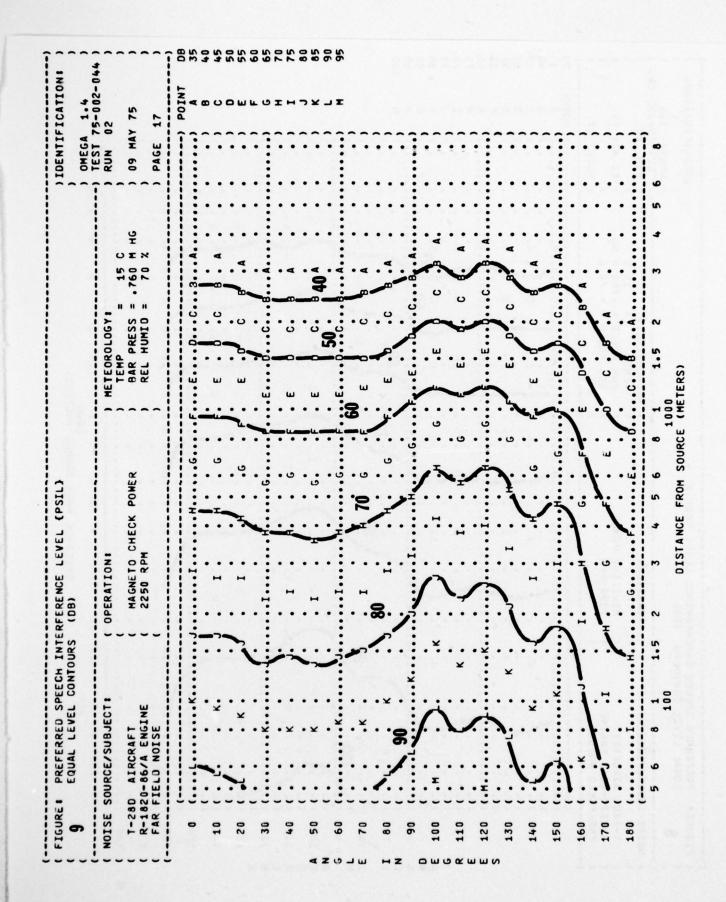


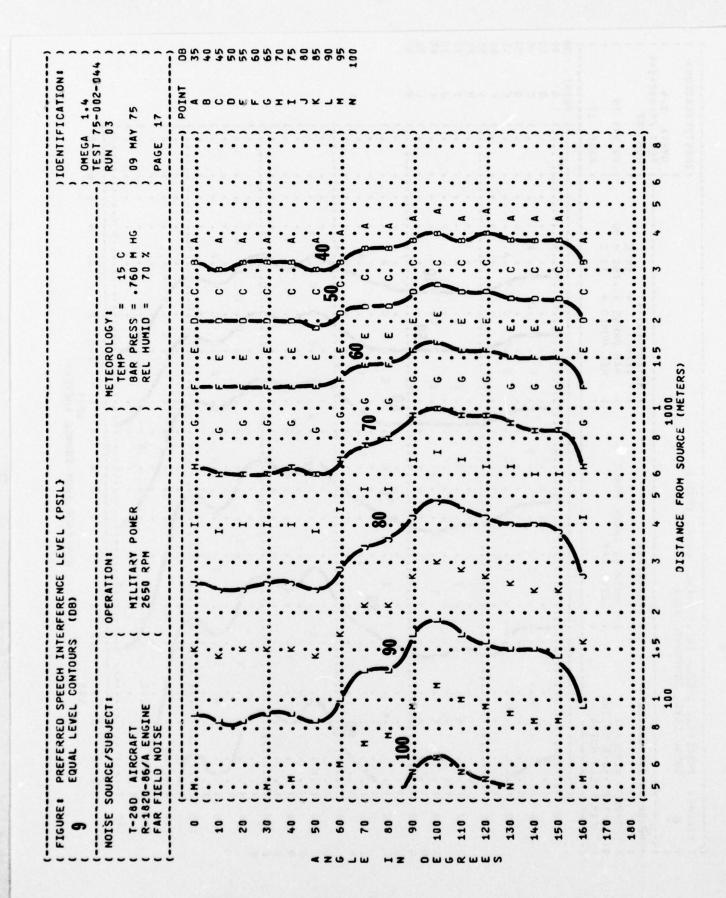












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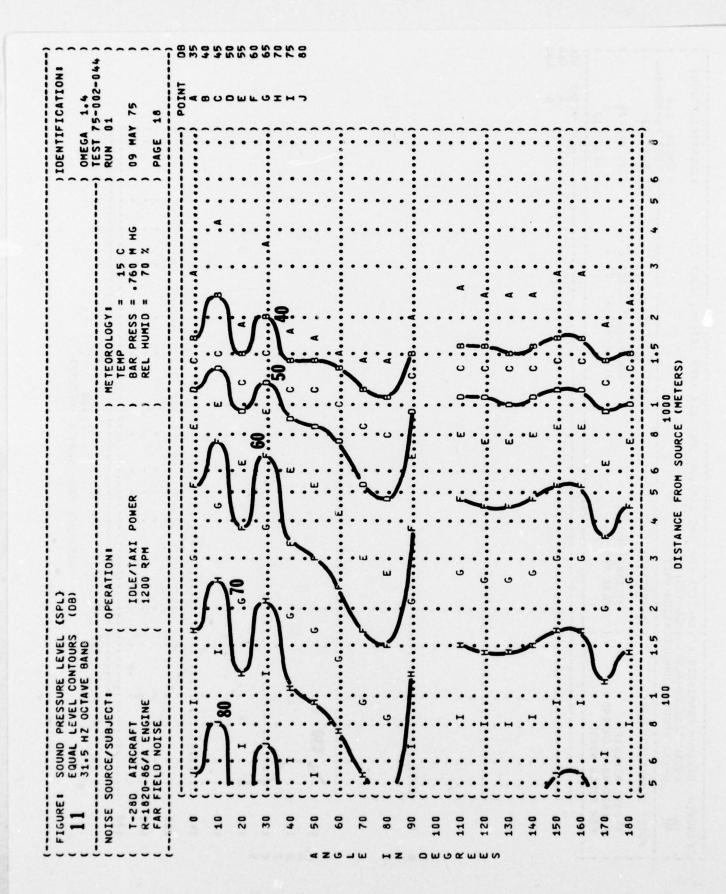
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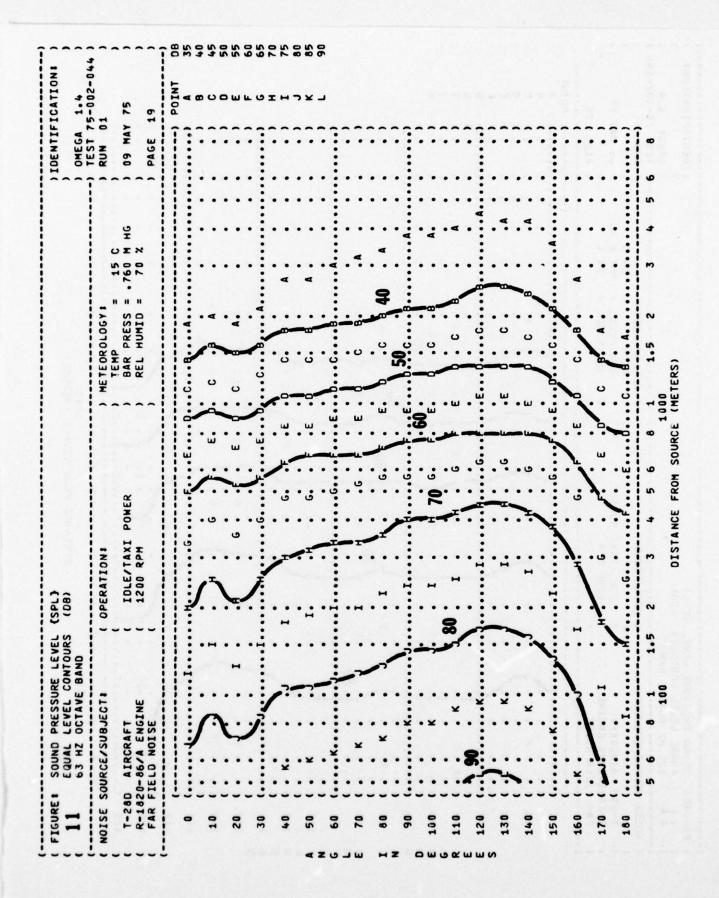
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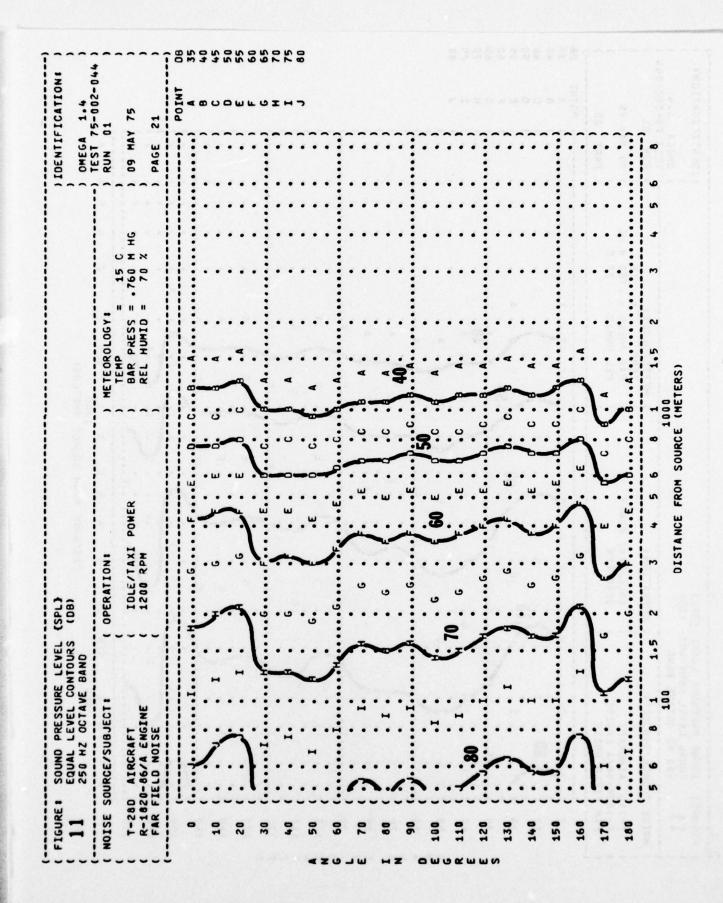
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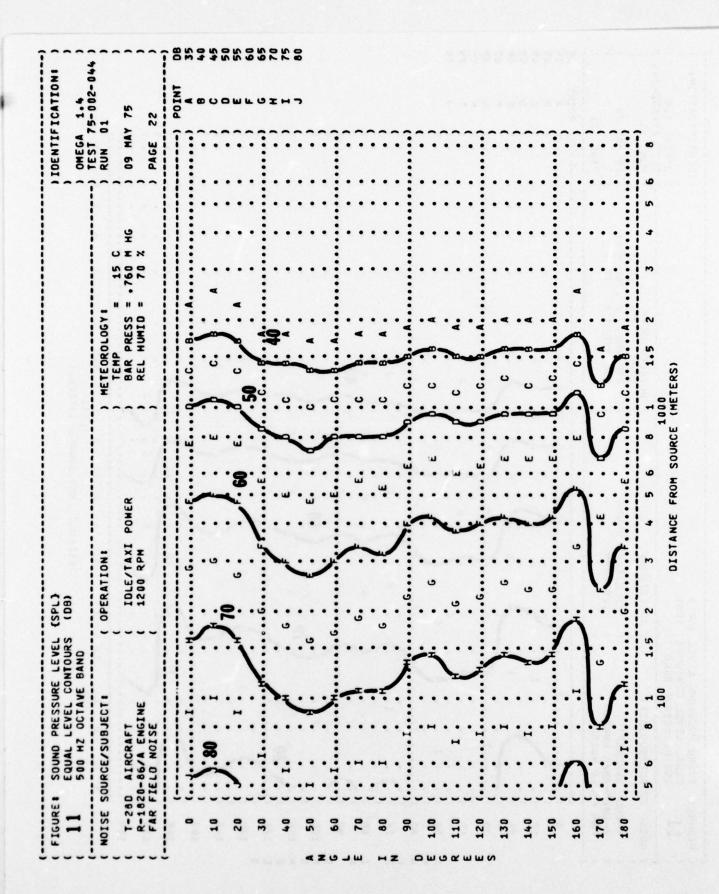
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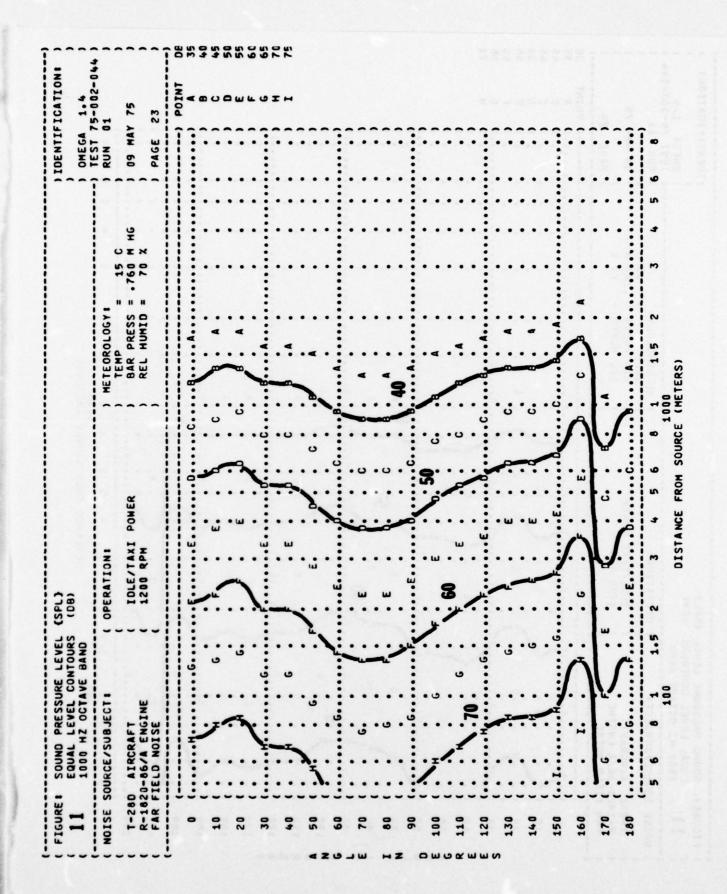




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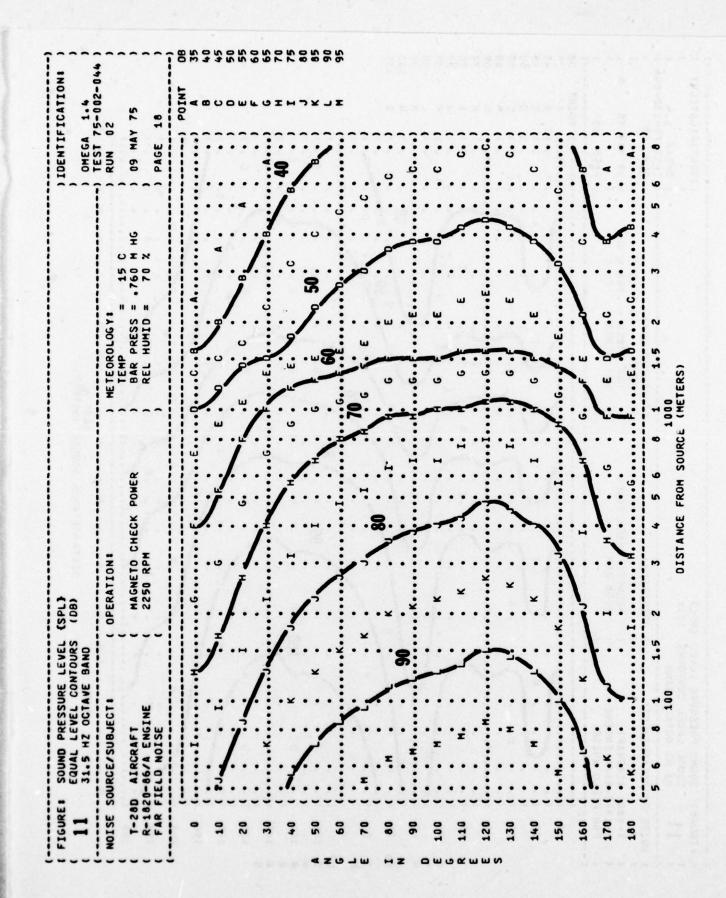


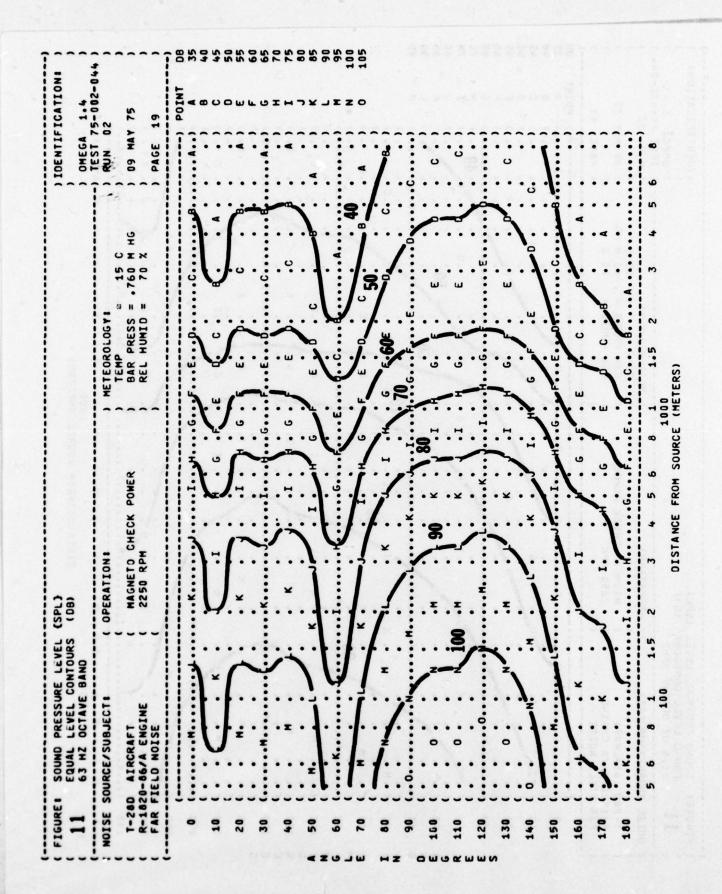


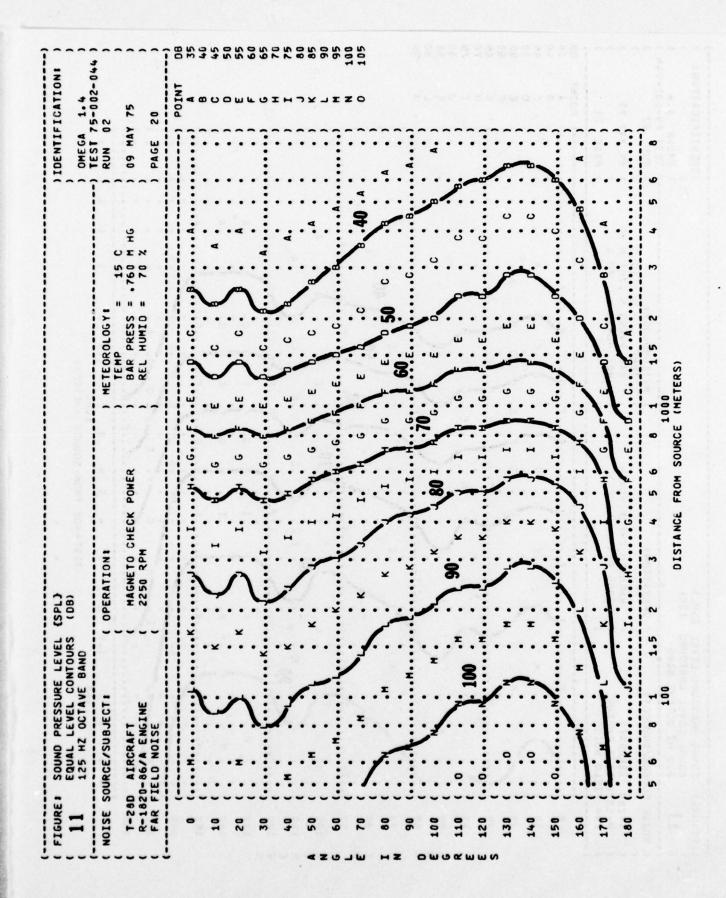
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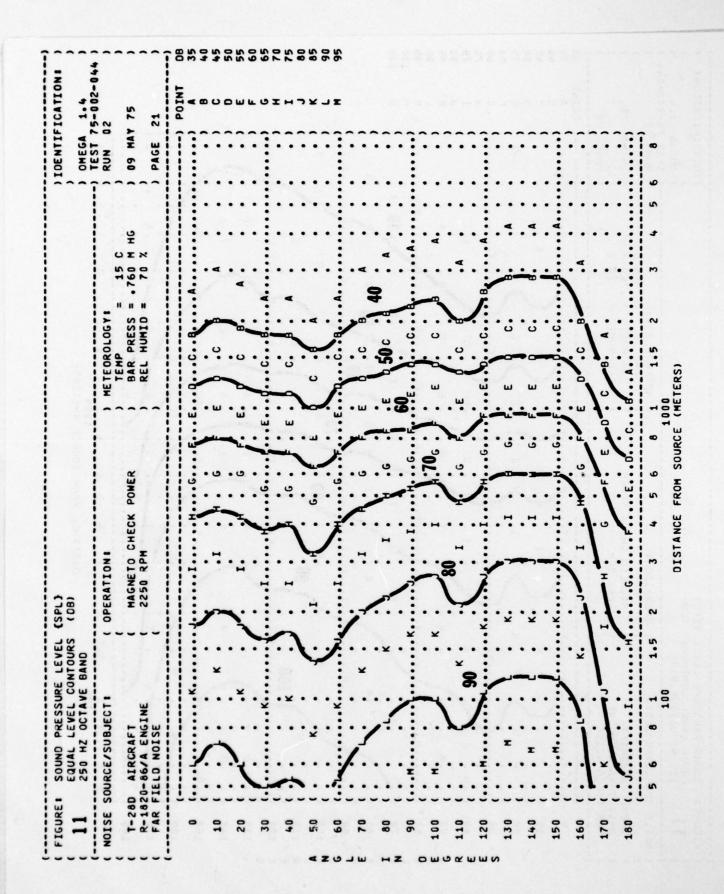
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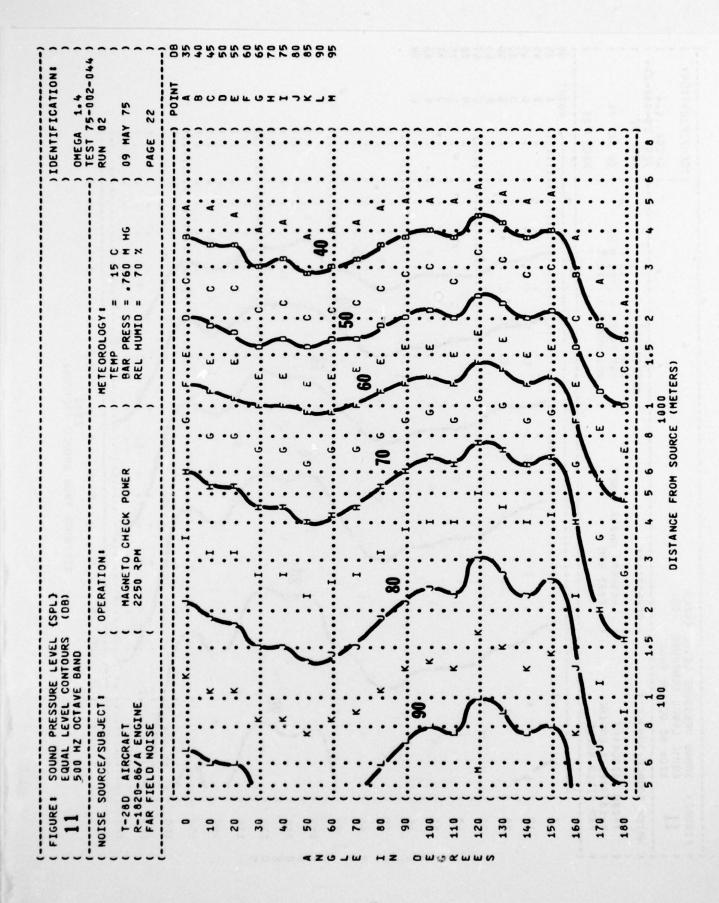
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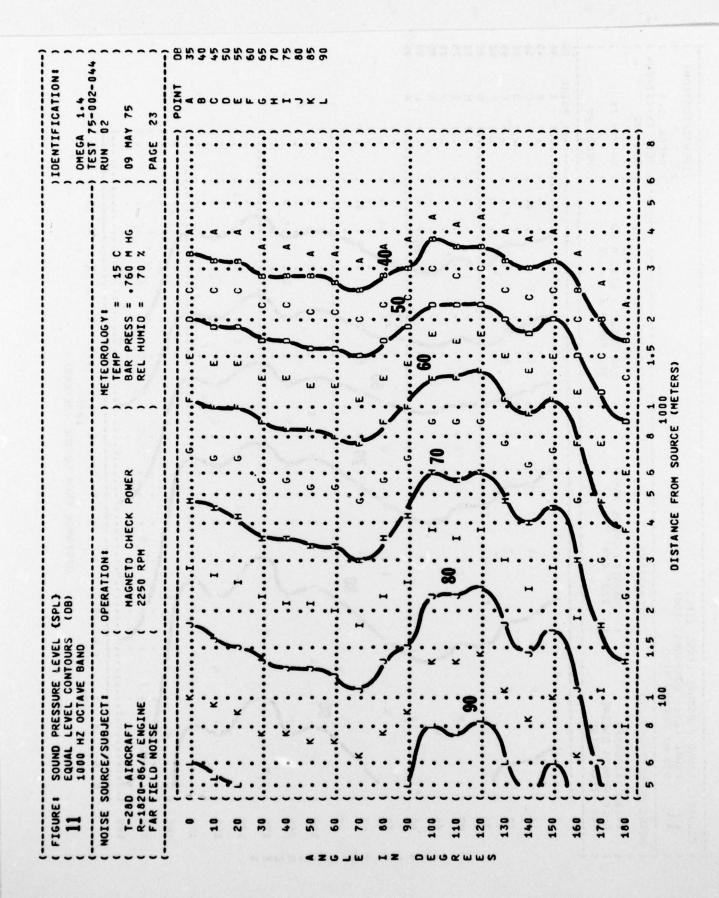


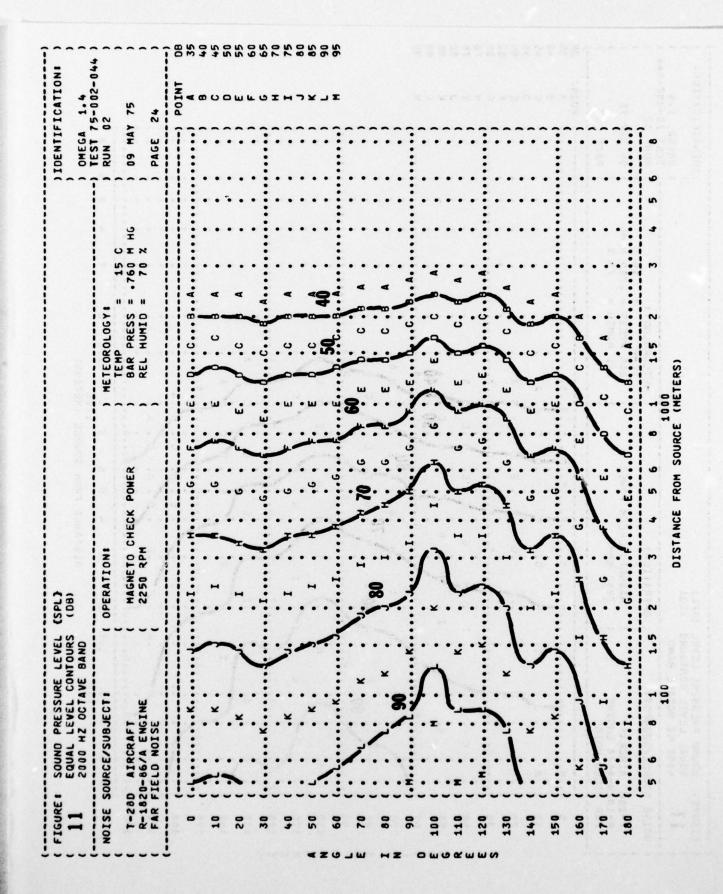


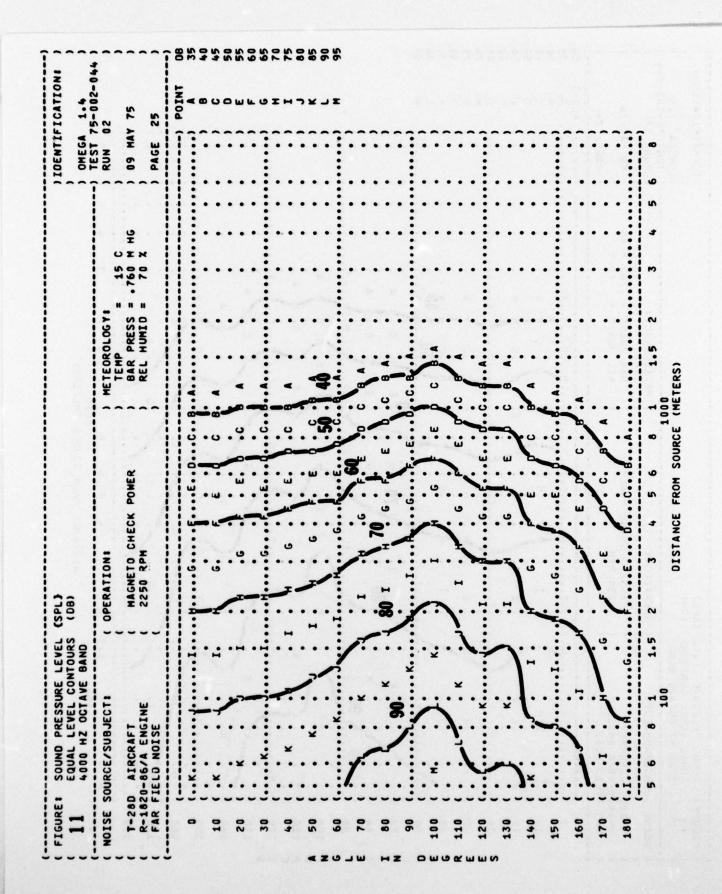












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